# Dynamic Response of Submerged Vertical Cylinder with Lumped Mass under Seismic Excitation 

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## PAPER INFO

## Paper history:

Received 10January2014
Received in revised form15 June 2014
Accepted 26 June 2014

## Keywords:

Vertical Circular Cylinder
Seismic Excitation
Lumped Mass
Added Mass

## $A B S T R A C T$

An analytical approach is presented to assess the response of offshore structures under seismic excitation. This paper evaluates the impacts of different fluid field models and the mass of equipment at the top of offshore structure which is simulated as lumped mass on the responses of offshore structures. To do this, two and three dimensional fluid field models are developed. In three dimensional models different approximation regarding the free surface boundary condition associated with high and low frequency excitations are adopted. Then the alternation of response of structure with changing in the value of lumped mass is calculated. Finally the impacts of different models on the value of maximum displacement for Kobe earthquake are evaluated. It is shown that different approximations regarding the fluid field could largely change the value of maximum displacement evaluated by the models.
doi: 10.5829/idosi.ije.2014.27.10a.08

## 1. INTRODUCTION

For maximum water depth less than 1000 m , it is usual to use platforms fixed in the bottom. Fixed bottom platforms include different kinds ranging from jackup rigs, gravity platforms, jacket platforms to compliant tower platforms. Considering fixed platforms, equipment mass aresimulated as lumped mass at the top of offshore structure. The surrounding fluid could have large impact on overall response of these structures when they are excited by earthquakes. The mathematical derivation of fluid-structure interaction requires some approximation on the fluid field that could affect the magnitude of the response estimated by the model.

Yeung [1] found the added mass and damping of floating circular cylinder in finite depth water. He derived added mass and damping for heave, sway and roll motions of the cylinder. Rahman et al. [2] obtained

[^0]added mass and damping coefficients for a circular cylinder clamped at the seabed. They used eigenfunction expansion including propagating and evanescent modes and finally developed asymptotic high and low frequency solutions for added mass and added damping. Williams [3] investigated dynamic response of surface piercing clamped circular cylinder under horizontal ground excitation. Employing high frequency approximation of free surface boundary condition, he developed Green function to obtain integral solution for compressible fluid flow. Tung [4] studied the behavior of submerged vertical cylindrical tank under harmonic ground excitations. Assuming incompressible, inviscid and irrotational fluid flow, Laplace equation was solved for two inner and outer regions of the fluid field and then, applying boundary and continuity conditions, hydrodynamic forces were obtained and effect of tank geometry properties on the hydrodynamic forces in graphical form were presented. Maheri et al. [5] experimentally calculated the added mass for rigid and flexible cylinders. Han et al. [6] derived an analytical solution for added mass of flexible cylinder under harmonic ground motion
excitation. They also developed a simple formula for computing natural frequency of the coupled system. Hsi-teh [7] investigated earthquake response of circular column for partially submerged water. The added mass for submerged floating tunnel in the case of deep and shallow water was derived by Shahmardani et al.[8].

Anagnostopoulos [9]who studied dynamic response of offshore platforms under wave loading used a time domain solution using Morrison's equation to account for fluid-structure interaction. Lee et al. [10] investigated the seismic response of flexible underwater oil storage tanks under horizontal ground motion. Bhata et al. [11] investigated dynamic response of vertical circular cylinder under wave loads with small amplitude. They considered three motions, surge, heave and pitch in finite-depth water and assuming four velocity potentials for water field, utilized an analytical solution for solution of fluidstructure interaction. Wu et al. [12] obtained exact solutions for natural frequencies and mode shapes of an immersed wedge beam with a tip mass at the end. They found good agreement between their results and finite element method. Oz [13] used analytical and finite element methods to calculate natural frequency of a beam partially immersed in the water with a tip mass. He considered transverse vibration of beam and showed that by increase in water height, tip mass and water density, there is a decrease in natural vibration frequencies. Naghipour [14] described various timedomain methods useful for analyzing the experimental data obtained from a circular cylinder force in terms of both wave and current for estimation of the drag and inertia coefficients applicable to the Morison's equation.

Esper [15] investigated the seismic response of an offshore structure using ANSYS. He used Westergaard added mass model to account for fluid structure interaction. He concluded that accounting for inertial effect of surrounding water reduces the maximum acceleration at platform level. Amundsen [16] investigated the influence of different models regarding fluid-structure interaction on the overall seismic response of a concrete platform. He considered two models, the first is an added mass model employing Morrison's equation and the second involves coupled analysis of the fluid and structure fields in ABAQUS. He found that although the change in the period of the system in the two models is negligible, the frequency content of the responses could have appreciable differences. The effect of added mass fluctuation on vertical vibration of TLP in the case of vibration in still water for both free and forced vibration subjected to axial load at the top of the leg was presented by Tabeshpour et al. [17].

Accounting for fluid-structure interaction through a complex coupled field analyses is numerically expansive. On the other hand, oversimplification of
this interaction through commonly adopted Morrison's equation, which does not account for reduced added mass near free surface, could significantly affect the accuracy of the analysis. This shows the need for analytical models capable of reproducing global response of the coupled field with good accuracy.

In this study, dynamic response of vertical circular cylinder with lumped mass is investigated under seismic excitation. Two and three dimensional models are adopted to consider fluid-structure interaction. Then governing equation of the system under simultaneous application of the fluid pressure and seismic excitation is obtained. Finally, variations of maximum displacement and natural frequencies of the coupled system with change in the ratio of lumped mass to the cylinder mass are investigated.

## 2. FORMULATION FOR JACKET

Offshore structure is simulated as a vertical circular cylinder with a lumped mass which is the mass of equipment at the top of it. As it has been shown in Figure 1, height, radius and thickness of cylinder are denoted by $h_{1}, r$ and $t$, respectively; and lumped mass $M$ is located at the top of cylinder. In this study the only rotary inertia of the lumped mass is ignored. The equilibrium equation of the cylinder with surrounding water is (Williams and Moubayed [18]):
$m \frac{\partial^{2} w_{t}}{\partial t^{2}}+E I \frac{\partial^{4} w}{\partial z^{4}}=m \ddot{w}_{t}+E I w^{(i v)}=F$
where $m$ and $E I$ are mass per unit length of cylinder and flexural rigidity of cylinder. Decomposing the deflection of the cylinder we have:
$w_{t}(z, t)=w(z, t)+w_{g}(g)$
here $w_{i}$ and $w$ are total and relative (to ground) lateral deflection of cylinder and $w_{g}$ is ground displacement. In Equation 1, $F$ simulates the external fluid force on cylinder body. The boundary conditions of cylinder are (To [19]):
$\left.w\right|_{z=0}=0$
$\left.\frac{\partial w}{\partial z}\right|_{z=0}=0$
$\left.E I \frac{\partial^{2} w}{\partial z^{2}}\right|_{z=h}=0$
$\left.E I \frac{\partial^{3} w}{\partial z^{3}}\right|_{z=h}=\left.M \frac{\partial^{2} w}{\partial t^{2}}\right|_{z=h}$
Considering free vibration of beam ( $F=0$ in Equation 1) with tip mass (boundary conditions of Equation 3), we could derive the eigen-functions (mode shapes) of
the beam. Now, employing Equation (2), we could expand total lateral deflection in terms of eigenfunctions of beam with tip mass as [19]:
$w_{t}(z, t)=\sum_{n=1}^{\infty} A_{n}(t)$
$\left[\sinh \varepsilon_{n} \frac{z}{h}-\sin \varepsilon_{n} \frac{z}{h}-\frac{\sin \varepsilon_{n}+\sinh \varepsilon_{n}}{\cos \varepsilon_{n}+\cosh \varepsilon_{n}}\left(\cosh \varepsilon_{n} \frac{z}{h}-\cos \varepsilon_{n} \frac{z}{h}\right)\right]$
$+w_{g}(t)=\sum_{n=1}^{\infty} A_{n}(t) \phi_{n}(z)+w_{g}(t)$
where $A_{n}(t)$ is modal amplitude of the $n^{\text {th }}$ mode and $\varepsilon_{n}$ is corresponding eigen-value. The eigen-value is determined solving the following equation [19]:
$\varepsilon_{n} \frac{\sin \varepsilon_{n} \cosh \varepsilon_{n}-\sinh \varepsilon_{n} \cos \varepsilon_{n}}{1+\cos \varepsilon_{n} \cosh \varepsilon_{n}}=\frac{m h}{M}=\frac{1}{\alpha}$
Hereafter we define $\alpha$ as mass ratio. As it is shown in Figure 2, increasing the ratio of lumped mass to mass per length of cylinder ( $M / \mathrm{mh}$ or mass ratio) there is decrease in the value of $\varepsilon_{n}$, which these decreasesare smaller for higher mode numbers. On the other hand, as it could be inferred from this figure, for larger mass ratios the change in the value of eigenvalue decreases and they become essentially constant.

## 3. FORMULATION FOR FLUID FIELD

Accounting for fluid-structure interaction through a complex coupled field analyses is numerically expansive. On the other hand, oversimplification of this interaction through commonly adopted Morrison's equation does not account for reduced added mass near free surface and consequently could affect the accuracy of the analysis. This shows the need for numerical inexpensive analytical models capable of reproducing global response of the coupled field with good accuracy.Assuming that the surrounding fluid is incompressible, inviscid and for irrotational fluid flow, it is possible to treat the flow field by potential theory. With this assumptions, the velocity potential function should satisfy Laplace equation:

$$
\begin{equation*}
\nabla^{2} \phi=0 \tag{6}
\end{equation*}
$$

Here we adopt two approaches to evaluate the fluid pressure on the cylinder body. In the first approach, the fluid field is treated as two dimensional (2D) field, assuming constant inertial effect of the fluid field along cylinder length (ignoring free surface boundary condition). In the second approach, the fluid field istreated as three dimensional (3D) field to account for the change in the inertial contribution of the fluid field along the cylinder length (especially near free surface).

The 3D models are developed employing simplification of free surface boundary condition. Two 3D models are developed which are only applicable for


Figure 1. Schematic view of vertical circular cylinder


Figure 2. Variations of eigenvalues with changing in the mass ratio
excitation with high or low frequency content. These analytical models could provide an upper and lower boundaries for displacement of the offshore.The general free surface boundary condition has the following form:

$$
\begin{equation*}
\frac{\partial^{2} \varphi}{\partial t^{2}}+g \frac{\partial \varphi}{\partial z}=0 \tag{7}
\end{equation*}
$$

To derive the simplified free surface boundary condition for high and low frequency excitation, we consider harmonic excitation in the form of $e^{i \omega t}$. For this excitation we have:
$\varphi(r, \theta, z, t)=e^{i \omega t} \varphi(r, \theta, z)$
now the boundary condition (7) becomes:
$\frac{\partial^{2} \varphi}{\partial t^{2}}+g \frac{\partial \varphi}{\partial z}=\left(-\omega^{2} \varphi(r, \theta, z)+g \frac{\partial \varphi(r, \theta, z)}{\partial z}\right) e^{i \omega t}$
and consequently:
$-\omega^{2} \varphi_{1}(r, \theta, z)+g \frac{\partial \varphi(r, \theta, z)}{\partial z}=0$
Considering low frequency excitation, this leads to:
$\frac{\partial \varphi(r, \theta, z)}{\partial z}=0$
and for high frequency excitation to:

$$
\begin{equation*}
\varphi(r, \theta, z)=0 \tag{12}
\end{equation*}
$$

In the following subsections, the solutions for two and three dimensional fluid fields are presented.
3. 1. Two Dimensional Fluid Field For two dimensional case, the Laplace Equation (6) becomes:
$\nabla^{2} \varphi=\frac{\partial^{2} \varphi}{\partial r^{2}}+\frac{1}{r} \frac{\partial \varphi}{\partial r}+\frac{1}{r^{2}} \frac{\partial^{2} \varphi}{\partial \theta^{2}}=0$
$R \leq r, \quad 0 \leq \theta \leq 2 \pi$
The potential function must also satisfy the following boundary conditions:
$\lim _{r \rightarrow \infty} \varphi=0$
$\left.\frac{\partial \varphi}{\partial r}\right|_{r=R}=\dot{w}_{t} \cos \theta$
The first boundary condition accounts for still water in large distance from the jacket, and the second boundary condition satisfies the compatibility of deflection in the fluid-structure interface. Imposing the boundary conditions, the velocity potential function is obtained in the following form:
$\varphi(r, \theta)=-\dot{w}_{t} \frac{R^{2}}{r} \cos \theta$
Integrating the fluid pressure on wet surface of the cylinder, the fluid force will be:
$F=-\rho_{f} \pi R^{2} \ddot{w}_{t}=-m_{a} \ddot{w}_{t}$
Now introducing this fluid force in Equation (1), we have:

$$
\begin{equation*}
\left(m+m_{a}\right) \ddot{w}_{t}+E I W^{(i v)}=0 \tag{17}
\end{equation*}
$$

Substituting $w_{t}$ from Equation (4), multiplying both sides of equation by $\varphi_{i}$ and integrating through the length of the beam and finally employing the orthogonality of cylinder's mode shapes, the differential equation for modal amplitudes could be derived:
$\left(m+m_{a}\right) \ddot{A}_{i}+E I \varepsilon_{i}^{4} A_{i}(t)=-\frac{2}{L} F_{i}$
$F_{i}=\left(m+m_{a}\right) \int_{0}^{h} \phi_{i}(z) \ddot{w}_{g}(t) d z$
3. 2. Three Dimensional Fluid Field The threedimensional model accounts for the change in the inertial contribution of the fluid along the cylinder length. In this section the external force is calculated for two different conditions, imposing different assumptions regarding the free surface boundary
condition. In this case the Laplace equation takes the following form:
$\nabla^{2} \varphi=\frac{\partial^{2} \varphi}{\partial r^{2}}+\frac{1}{r} \frac{\partial \varphi}{\partial r}+\frac{1}{r^{2}} \frac{\partial^{2} \varphi}{\partial \theta^{2}}+\frac{\partial^{2} \varphi}{\partial z^{2}}=0$
subjected to the following boundary conditions:
$\varphi=0 \quad$ as $: r \rightarrow \infty$
$\frac{\partial \varphi}{\partial r}=\dot{w}_{t} \cos \theta \quad$ at $: r=R$
$\frac{\partial \varphi}{\partial z}=0 \quad$ at $: \quad z=0$
The first equation accounts for still water at large distance from jacket, and the second equation considers the compatibility of deflection between jacket and its surrounding water. The last equation is the impermeability boundary condition at bottom (no fluid flow (velocity) in the direction normal to sea bed).
We derive the asymptotic solution for two cases of near zero and high frequency excitations. To do this, two 3D solution accounting for two simplified boundary conditions for free surface (Equations (11) and (12)) are considered.

Considering Equation (12), in the case of high frequency excitation (solution denoted by $\varphi_{1}$ ), the free surface boundary condition simplifies to:
$\varphi_{1}=0 \quad$ at $: z=h$
In the same time, considering Equation (11), for near zero (low) frequency excitation (solution denoted by $\varphi_{2}$ ), the free surface boundary condition reduces to:
$\frac{\partial \varphi_{2}}{\partial z}=0 \quad$ at: $z=h$
Imposing the boundary conditions, the analytical solution for the potential function are obtained as:
$\varphi_{1}(r, \theta, z, t)=\sum_{m=1}^{\infty} B_{m}(t) K_{1}\left(\lambda_{m} r\right) \cos \left(\lambda_{m} z\right) \cos \theta$
$\lambda_{m}=\frac{(2 m-1) \pi}{2 h}$
and
$\varphi_{2}(r, \theta, z, t)=\sum_{m=1}^{\infty} A_{m}(t) K_{1}\left(\beta_{m} r\right) \cos \left(\beta_{m} z\right) \cos \theta$
$\beta_{m}=\frac{m \pi}{h}$
where $K_{l}$ is the modified Bessel function of second kind of order one and $\lambda_{m}$ and $\beta_{m}$ are the problem eigenvalues. The only difference between two solutions is in the value of eigenvalues. To avoid repetitive derivations, we only derive the solution for $\varphi_{1}$.

Considering the compatibility condition at fluidstructure interface, we conclude:
$\sum_{m=1}^{\infty} B_{m}(t) \lambda_{m} K_{1}^{\prime}\left(\lambda_{m} R\right) \cos \left(\lambda_{m} z\right) \cos (\theta)$
$=\sum_{n=1}^{\infty}\left[\phi_{n}(z) A_{n}(t)+\dot{w}_{g}\right] \cos (\theta)$
Now making use of the orthogonality of harmonic functions, we have:
$B_{i}(t)=\frac{2}{h \lambda_{i} K_{1}^{\prime}\left(\lambda_{i} R\right)} \sum_{n=1}^{\infty}\left[\gamma_{n i} A_{n}(t)+\gamma_{i} \dot{W}_{g}(t)\right]$
where:

$$
\begin{align*}
& \gamma_{n i}=\int_{0}^{h} \cos \left(\lambda_{i} z\right) \phi_{n}(z) d z  \tag{27}\\
& \gamma_{n}=\int_{0}^{h} \cos \left(\lambda_{i} z\right) d z
\end{align*}
$$

Thus the potential function takes the following form:
$\varphi_{1}(r, \theta, z, t)=\sum_{m=1}^{\infty} \sum_{n=1}^{\infty} \frac{2 K_{1}\left(\lambda_{m} r\right)}{h \lambda_{m} K_{1}^{\prime}\left(\lambda_{m} R\right)}$
$\left[\gamma_{n m} \dot{A}_{n}+\gamma_{m} \dot{W}_{g}(t)\right] \cos \left(\lambda_{m} z\right) \cos (\theta)$
Integrating the fluid pressure over the cylinder wet surface, the fluid induced inertial force imposed on the structure could be evaluated as:
$F_{1}=-m_{a} \sum_{m=1}^{\infty} \sum_{n=1}^{\infty}\left[\gamma_{n m} \ddot{A}_{n}+\gamma_{m} \ddot{W}_{g}(t)\right] C_{m} \cos \left(\lambda_{m} z\right)$
where
$C_{m}=\frac{2 K\left(\lambda_{m} R\right)}{h \lambda_{m} R K^{\prime}\left(\lambda_{m} R\right)}$
Introducing this fluid force in structure's equilibrium equation (Equation (1)), we have:

$$
\begin{align*}
& \sum_{m=1}^{\infty} E I \phi_{m}^{i V}(z) A_{m}(t)+m \sum_{m=1}^{\infty} \phi_{m}(z) \ddot{A}_{m}(t)+m \ddot{W}_{g}(t) \\
& +m_{a} \sum_{m=1}^{\infty} \sum_{n=1}^{\infty}\left[\gamma_{n m} \ddot{A}_{n}+\gamma_{m} \ddot{W}_{g}(t)\right] C_{m} \cos \left(\lambda_{m} z\right)=0 \tag{31}
\end{align*}
$$

Multiplying both sides of Equation (31) by $\phi_{i}(z)$ and integrating over 0 to $h$, we obtain a set of coupled ordinary differential equations:
$E I\left(\frac{\varepsilon_{i}}{h}\right)^{4} A_{i}(t)+m \ddot{A}_{i}(t)+m_{a} \sum_{m=1}^{\infty} \sum_{n=1}^{\infty} C_{m} \gamma_{m n} \frac{\gamma_{i m}}{D_{i m}} \ddot{A}_{n}(t)=-\frac{F_{s 1}}{D_{i m}}$
$F_{s 1}=\left[m_{a} \sum_{m=1}^{\infty} C_{m} \gamma_{m} \gamma_{i m}+m E_{i}\right] \ddot{W}_{g}(t)=\left[M_{F s 1}+m E_{i}\right] \ddot{W}_{g}(t)$
$D_{i m}=\int_{0}^{h} \phi_{i} \phi_{m} d z$
$E_{i}=\int_{0}^{h} \phi_{i}(z) d z$
The differential equation for modal amplitudes of $\varphi_{2}$ becomes:
$E I\left(\frac{\varepsilon_{i}}{h}\right)^{4} B_{i}(t)+m \ddot{B}_{i}(t)+m_{a} \sum_{m=1}^{\infty} \sum_{n=1}^{\infty} C_{m} \gamma_{m n} \frac{\gamma_{i m}}{D_{i m}} \ddot{B}_{n}(t)=-\frac{F_{s 2}}{D_{i m}}$
$F_{s 2}=\left[m_{a} \sum_{m=1}^{\infty} C_{m} \gamma_{m} \gamma_{i m}+m E_{i}\right] \ddot{w}_{g}(t)=\left[M_{F s 2}+m E_{i}\right] \ddot{w}_{g}(t)$
These sets of differential equations are solved using 4th order Runge-Kutta integration method. The added mass matrix will have following components:
$m_{i n}=m_{a} \sum_{m=1}^{\infty} C_{m} \gamma_{m i} \frac{\gamma_{i m}}{D_{i m}}$
Existence of off-diagonal terms in this matrix shows the coupling of different modes in the differential equation for modal amplitudes.

In Equations (32) and (33), $F_{s 1}$ and $F_{s 2}$ represent the earthquake induced lateral force along the cylinder external surface. These forces are composed of two terms. The first term is due to fluid flow adjacent to the cylinder body and the second term is associated with the cylinder's own inertia. Note that difference in the value of $F_{s 1}$ and $F_{s 2}$ comes from the difference in the value of eigenvalues $\lambda_{m}$ and $\beta_{m}$. Figure 3 depicts the variation of $M_{F s 1}$ and $M_{F s 2}$ (masses associated with $F_{s 1}$ and $F_{s 2}$ ) with mass ratio and mode number. Interesting point is that while $M_{F s 1}$ (the mass associated with the fluid induced lateral force on the cylinder body in the case of high frequency excitation) is larger for decreasing mode numbers and increasing mass ratios, $M_{F s 2}$ (the mass associated with the fluid induced lateral force on the cylinder body in the case of low frequency excitation) becomes zero for all mass ratios and mode numbers. This shows that in the case of low frequency excitation, the lateral force along cylinder body comes only from the mass inertia of cylinder and there is no contribution from surrounding fluid.

Figure 4 shows the evolution of three dimensional added mass matrix for the cases of high and low frequency excitations when lumped mass is equal to zero. As it is evident from this figure, for both types of excitations diagonal terms are significantly larger than off-diagonal ones. This is an indication of negligible coupling between different modes in both cases.

Variations of diagonal terms of added mass matrix with changes in the mass ratio are shown in Figure 5. As it could be inferred from this figure, there is completely different pattern for low and high frequency excitations in the case of low mass ratios. For higher mass ratios in both cases diagonal terms of added mass matrix are insensitive to the change in the mass ratio.

Figure 6 illustrates a comparison of the diagonal terms of added mass matrix, obtained using twodimensional and three-dimensional models. It is shown that while in the 2 D model, the value of added mass remains constant for different modes, in the 3 D model, different modes have different ratios of added mass
which are less than the 2 D model. This reduction in the added mass of 3D model is due to imposition of impermeability and free surface boundary conditions in this model.


Figure 3. Variations of masses $M_{F S 1}$ and $M_{F s 2}$ with change in mass ratio and mode number, a) high frequency excitation and b) low frequency excitation


Figure 4. Value of different elements of added mass matrix for five modes in the case of zero lumped mass ( $M=0$ ), a) high frequency excitation and b) low frequency excitation


Figure 5. Evolution of diagonal terms of added mass matrix with mass ratio, a) high frequency excitation and b) low frequency excitation


Figure 6. Comparison of diagonal terms of added mass matrixes in 2D and 3D models

(a)

(b)

Figure 7. Kobe ground motion record, a) time history and b) Fourier decomposition.

## 4. GROUND MOTION EXCITATION

To evaluate the extent of the variation in response due to change in assumption regarding fluid field, in this section results of analysis of offshore structure for ground Kobe ground motion excitation are presented. The result shows how different models regarding fluid field could affect the evaluated response. By evaluating the response to Kobe ground motion record, it is not intended to obtain a global judgment about seismic response of the offshore structures. The main focus of the paper is on assessing the difference in the seismic response due to assumed models for fluid field.

Tabulated in Table 1 are the values of the parameters used in the analysis of reinforced concrete vertical cylinder. Analyses are done for ground motion record of Kobe earthquake ( 090 component of 1995 Kobe earthquake at Japan recorded at Nishi-Akashi station, KOBE/NIS000). In the following simulations the same value for h and h 1 is used. Table 2 shows duration, peak ground acceleration and epicentral distance of ground motion record. Figure 7 depicts time history of the ground motion record and its Fourier spectrum.

Figure 8 presents the variation in the natural frequencies of cylinder for different mass ratios for 3D model with high and low frequency excitations and 2D fluid field model, where $\omega_{1}$ and $\omega_{i}$ are first natural frequency of cylinder when mass ratio is equal to zero $(\alpha=0)$ and natural frequencies of cylinder for different added mass ratios, respectively. In all models the largest change in the frequency is for the first mode. For large mass ratios, which is applicable for most of platforms, 3D fluid model applicable for high frequency excitation results in lower frequencies in nearly all modes compared to the other models. This indicates that the participation of higher modes in 3D model applicable for higher frequencies could be higher compared to the other models and as expected this model could give better estimate of actual response for excitations rich in higher frequencies.

Evolution of maximum displacement at the top of structure with change in the mass ratio for four different cases, including 3D models with high and low frequency excitations, 2D model and ignoring fluid field (dry cylinder) are depicted in Figure 9. It can be seen that maximum displacement occurs when lumped mass is equal to zero and by increasing the value of lumped mass the maximum displacements decreases in all cases.

Interesting point is that maximum displacement in the case of 3D model with low excitation frequency is very close to those of dry cylinder. This could be due to zero value of Ms2 for all modes in the case of low frequency excitation. On the other hand, the results for 3D model with high frequency excitation tend to those of 2D model. However, note that there are appreciable differences in the estimations of the two models for maximum displacement.

TABLE 1. Parameters used in the analyses

| TABLE 1. Parameters used in the analyses |  |
| :---: | :---: |
| Parameter | Value |
| $\mathbf{E}\left(\mathbf{N} / \mathbf{m}^{2}\right)$ | $2.69 \times 1010$ |
| $\mathbf{h 1} \mathbf{h}(\mathbf{m})$ | 100 |
| $\mathbf{R}(\mathbf{m})$ | 5 |
| $\mathbf{t}(\mathbf{m})$ | 0.8 |

TABLE 2. Seismic data used in this study

| Input ground <br> motion | Duration <br> $(\mathbf{s})$ | PGA <br> $(\mathbf{g})$ | Epicentral distance <br> $(\mathbf{k m})$ |
| :---: | :---: | :---: | :---: |
| Kobe $(1995)$ | 40 | 0.4862 | 8.7 |


(b)

(c)

Figure 8. Variations of natural frequencies with change in lumped mass ratio, a) 3D fluid field with high frequency excitation, b) 3D fluid field with low frequency excitation and c) 2D fluid field.


Figure 9. Variations of maximum displacement with change in mass ratio.



Figure 10. Fourier decomposition of response for different models, a) 3D model with high frequency boundary condition, b) 3D model with low frequency boundary condition, c) 2D model and d) response without fluid.

The result also indicates that in the case of excitation rich in low frequencies, the evaluated maximum displacement employing 2D model will not be on the safe side.

Shown in Figure 10 is the Fourier decomposition of the response for mass ratio of one. As could be seen in all models main contribution comes from the first mode of response and the estimation of different fluid models from this frequency is essentially the same. This also shows that commonly adopted pushover analysis proportional to the first mode of response will give good approximation of actual response.

## 5. CONCLUSIONS

Response of vertical circular cylinder with a lumped mass at its top and surrounding fluid under ground motion has been investigated. Fluid force along the cylinder body is evaluated employing different approximations for fluid field. These include two dimensional fluid field and also three dimensional fluid field with different assumptions regarding free surface boundary conditions. Deriving the response for 3D models, the evolution of fluid force along the cylinder
body for different excitation frequencies is evaluated. It is shown that for low frequency excitation, the lateral force along the cylinder body comes only from the inertia of the cylinder body and there is no contribution from the surrounding fluid. The results also show that different approximations of the fluid field could have large impact on the value of calculated maximum displacement and in all models the main contribution comes from the first mode.

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## Dynamic Response of Submerged Vertical Cylinder with Lumped Mass under Seismic Excitation

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PAPER INFO

Paper history:
Received 10 January 2014
Received in revised form 15 June 2014
Accepted 26 June 2014

## Keywords:

Vertical Circular Cylinder
Seismic Excitation
Lumped Mass
Added Mass

يك روش تحليلى براى ارزيابى پاسخ سازههاى دريايى تحت اثر بار زلزله ارائه شده است. اين مقاله اثرات مدلهاى مختلف ميدان سيال و جرم تجهيزات در بالاى سازه دريايى را كه بصورت جرم متمركز مدل شده را بر روى پاسخ سازه دريايیى ارزيابى مى كند. مدل دو و سه بعدى ميدان سيال بسط داده شده است. درمدلهاى سه بعدى تقريبهاى مختلف با توجه به شرايط مرزى سطح آزاد در ارتباط با تحريى فركانس بالا و پايين اتخاذ شده است. سپس تغييرات پاسخ سازه همراه با تغيير در مقدار جرم متمركز محاسبه شده است. درنهايت اثرات مدلهاى مختلف به مقدار حداكثر جابجايى براى زمينلرزه كوبه ارزيابى شده
است. نشان داده شده است كه تقريبهاى مختلف در رابطه با ميدان سيال تا حد زيادى مىتواند مقدار حداكثر جابجايى با مدلهاى ارزيابى شده را تغيير دهد.
doi: 10.5829/idosi.ije.2014.27.10a.08


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